

CanSat took me to Earth orbits, Asteroids, and Beyond! -An experience of 1st-generation CanSat student

Institute of Space and Astronautical Science, JAXA Yuichi Tsuda

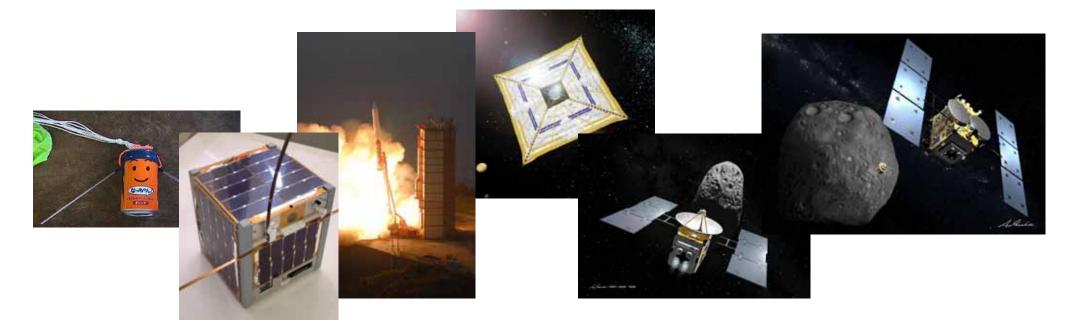
## Yuichi Tsuda

Professor, Institute of Space and Astronautical Science Japan Aerospace Exploration Agency

My field: Innovative spacecraft system, Solar system exploration, Astrodynamics

My history

- CanSat (1st generation!)
- CubeSat (1st generation!) Student project manager, U of Tokyo "XI-IV"
- IKAROS (World first deep space solar sail) *Deputy project manager*
- Hayabusa2 (Asteroid sample return mission) *Project Manager*



The beginning (1999~2003)

Journey begins suddenly...

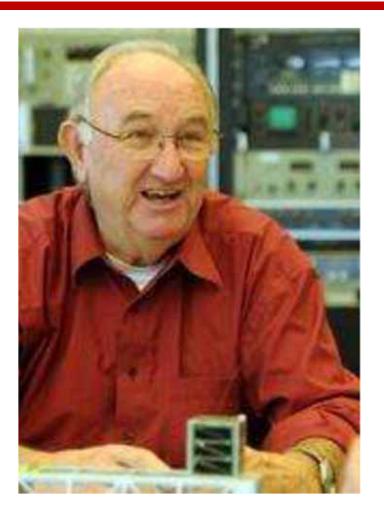
In 1998 when I was a student in Prof. Nakasuka's lab,

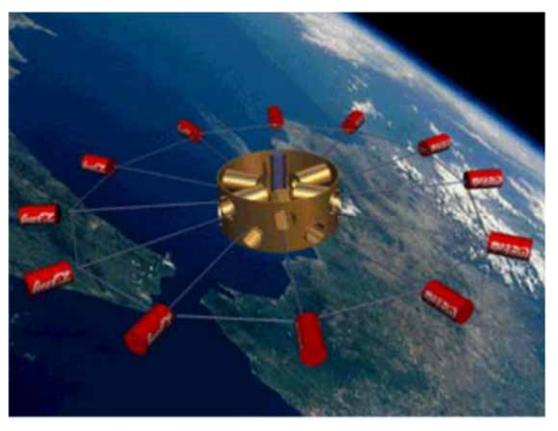
Professor: There is a workshop in Hawaii. Do you want to go?

> Me: Wow, Hawaii!? I will go. By the way what's that workshop for?

The workshop was "University Space Systems Symposium" where the CanSat was born.

#### Birth of "CanSat" Concept





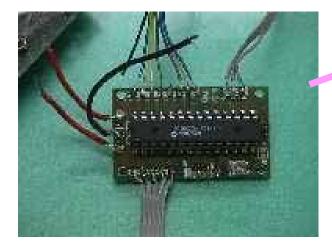
Initial Concept: launch all the CanSats and operate them in next USSS (one year later)

"Let's make a satellite out of this Coke-can !!" Prof. Bob Twiggs, Stanford University

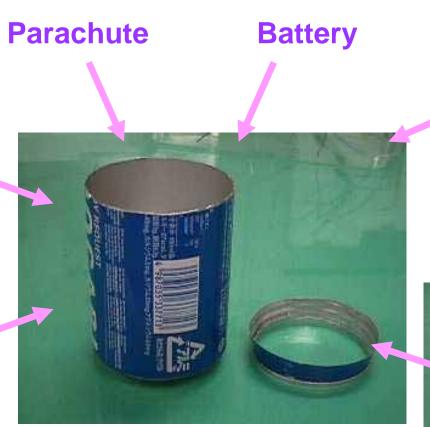
### First work



**Main Board** 

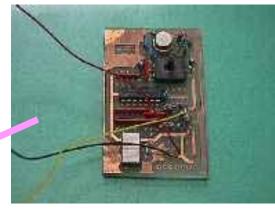


**Com encoder (TNC)** 

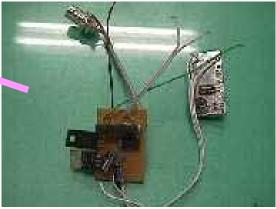


350ml Juice Can

**Antenna** 



#### **Sensors Board**



**Transmitter** 

#### 1st Generation "CanSats"

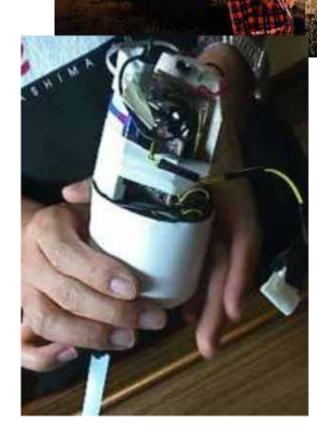


#### **Pre-final model**

#### Flight model of CanSat #003

Three 1st generation CanSats developed by U of Tokyo.

#### Launched in Black Rock Desert, 1999



### Test as you fly. Fly as you test.



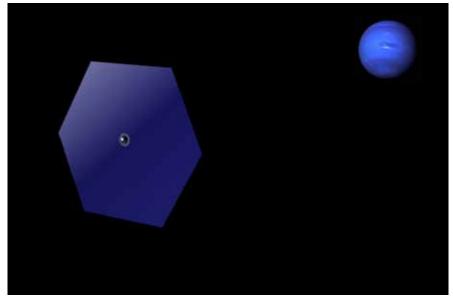


#### Want to make a Real Satellite ! ~ CanSat to CubeSat

Proposal from Prof. Twiggs again . Let's make a 10cm-cubic satellites!

Thin space membrane + CubeSat = ?





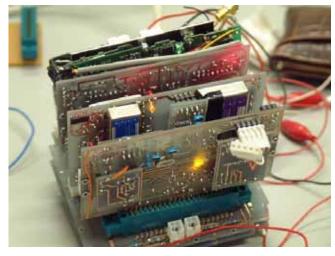
Ref. Tsuda, Nakasuka, Aoki, Nagashima, Space Transportation Symposium, 1999

1998 Pluto explorer concept with 200mdiameter thin flexible solar cells. How to fold it? "*Tsuda-folding*"
1999 We proposed "CubeSat + Space Membrane" concept, but finally resulted in...

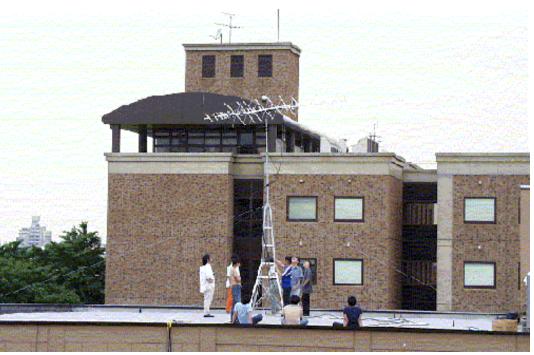
#### Textbook does not teach us.

The unexplored world should be pioneered by ourselves!









#### World First CubeSat "XI-IV"



## 2003/06/30 18:15:26 (local time) XI-IV was Launched!!!



#### CubeSat XI-IV Photo Gallery July – November 2003, University of Tokyo ISSL



### 7.30 South Atlantic 9.14 Azores Islands 9.17 East Timor







10.5 Bangladesh

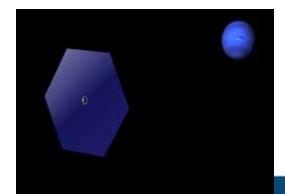
10.5 Tibet

11.03 Egypt

#### From CubeSats to Deep Space Exploration (2003 ~ )

#### "Continuity" in my first 10 years of carrier

(C)JAXA



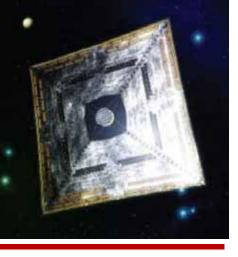
1998 Pluto explorer using space membrane technology (Concept study)

#### CanSat for "Furoshiki" Study

•Equipped with a thin flexible membrane around the satellite •Folded within the shape of can while launch •Verify the characteristics of the folding/extension/method •Estimate the shape of membrane 1999 CanSat membrane deployment experiment (Concept study + Development) → *CanSat w/o membrane* 

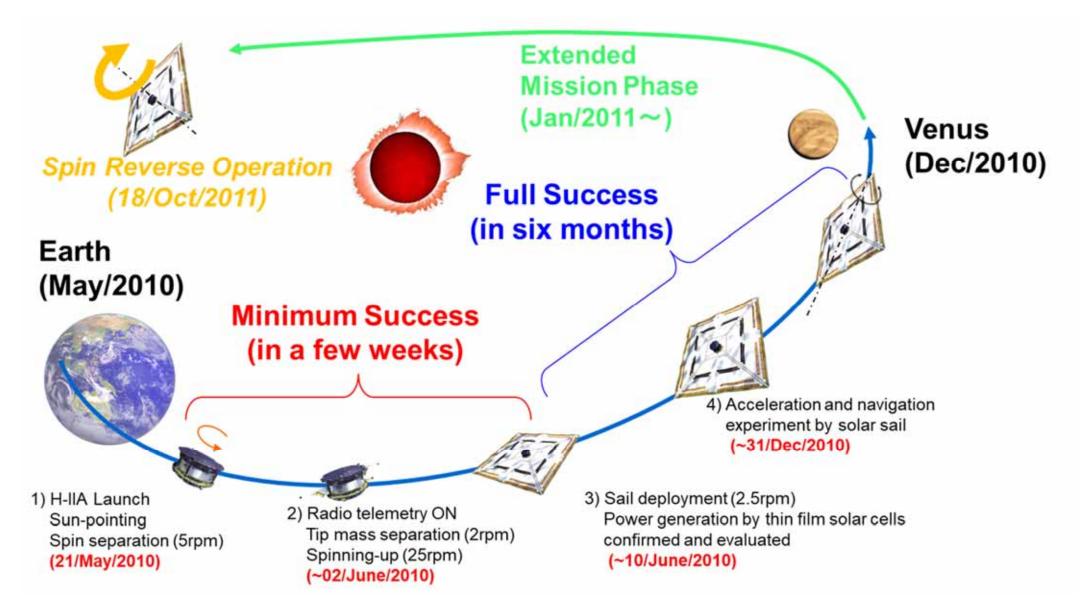
2000 CubeSat membrane deploymene experiment (Concept study + Development) → CubeSat w/o membrane

> 2007 IKAROS Solar Power Sail Demonstrator →*Real Space mission!*



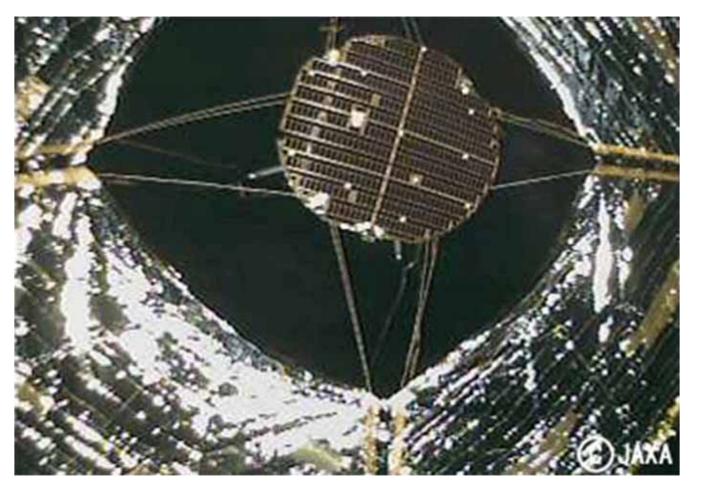
#### IKAROS (Interplanetary Kite-craft Accelerated by Radiation Of the Sun)







Self-photo to confirm the sail deployment in interplanetary field by interplanetary CanSat.





Jun 14, 2010.

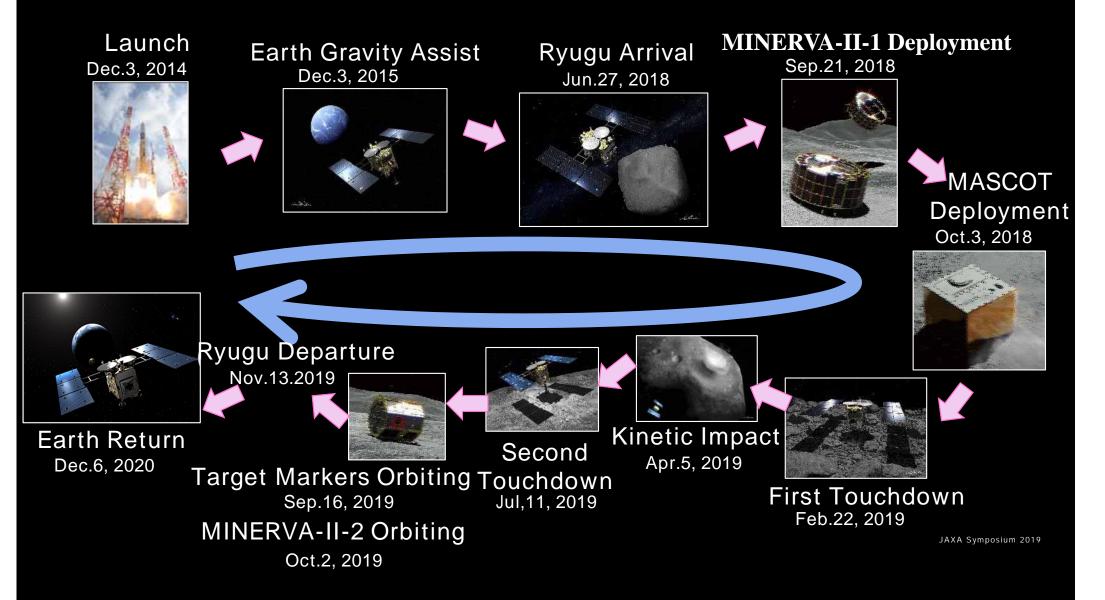
**Toward Hayabusa2** 



## Hayabusa2 Mission



#### Sample return mission to a C-type asteroid "Ryugu"



### Hayabusa2's mission history

2014年12月3日

「はやぶさ2」 打上げ

Launch

#### Earth Swinby Ion Engine Cruise Ryugu Arrival Minerva-II1 landing -Success Mascot landing -Success Touch down #1 -Success Kinetic Impact —Success Touch down #2 -Success Minerva-II2 orbiting/landing -Success Earth Return

Got Ryugu sample! —to Extended Mission

## Asteroid "Ryugu"

- Top shape with a very circular equatorial bulge
- Spectrum type: Cb (Carbon-rich)
- Diamter: ~ 900 m
- Mass: ~ 450 million ton
- Obliquity: ~8 °
- Rotation period : P = 7.63 hours
- Reflectance factor (v-band) : 0.02
- Terrain: Very bumpy



HAYABUSA2 ONC-T TIME: UT 2018/07/10 06:05:08 (C)JAXA, U of Tokyo, et.al.



### Accomplishments of Hayabusa2 (1/2)



CAM-H image at the 1st touchdown

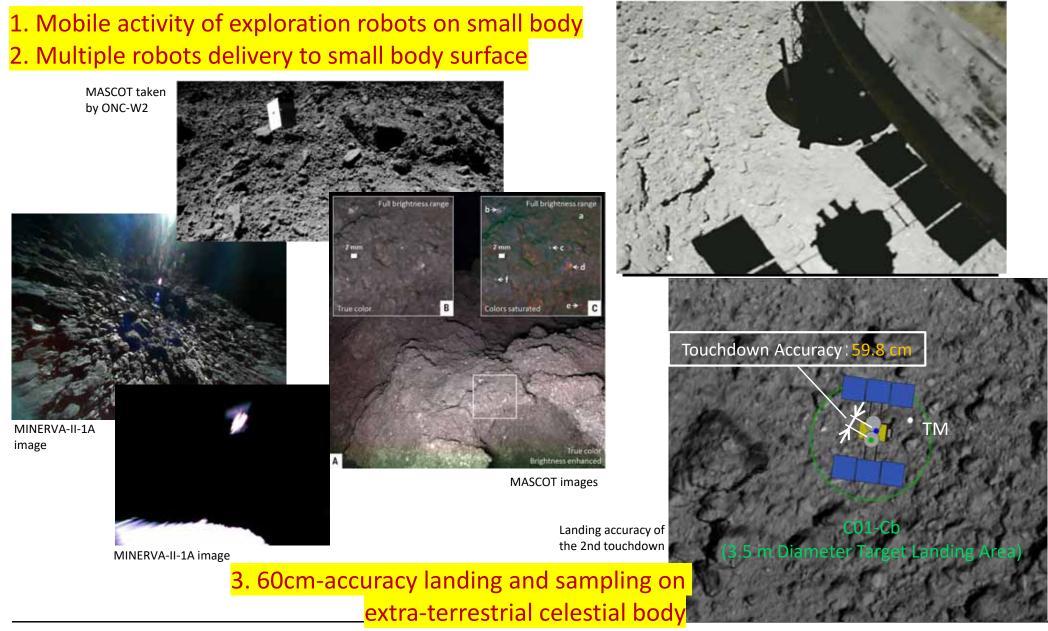


image credit: JAXA



### Accomplishments of Hayabusa2 (2/2)



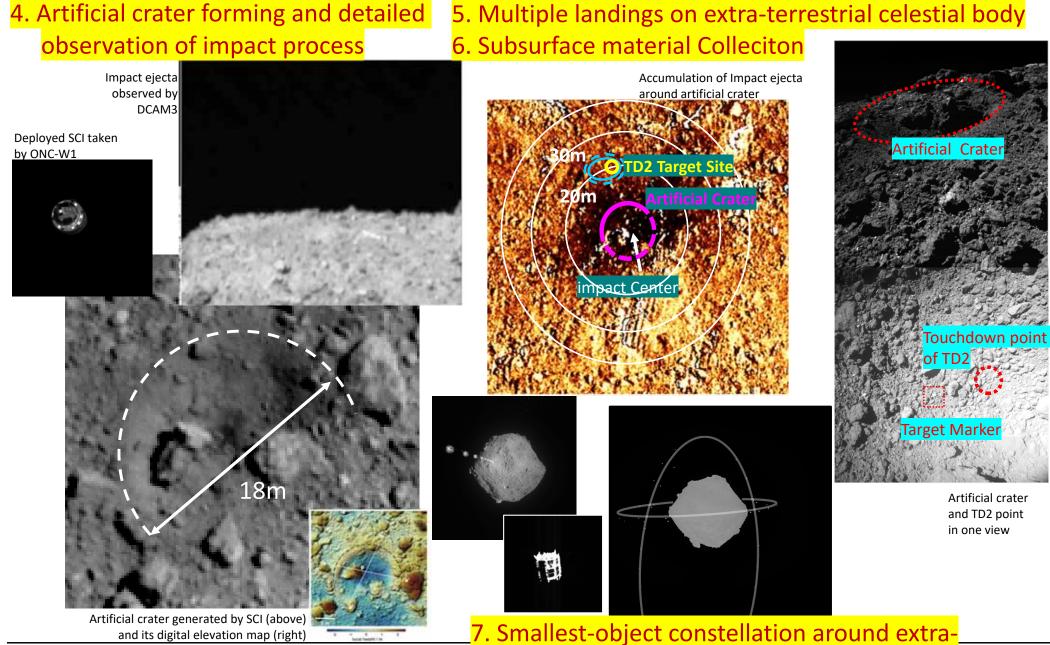
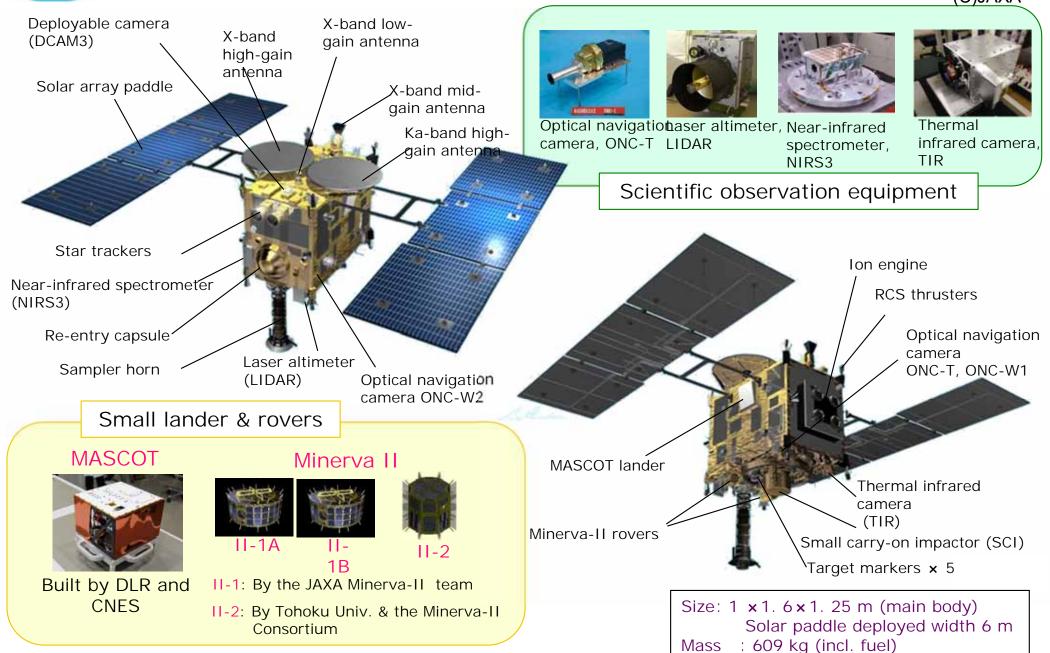


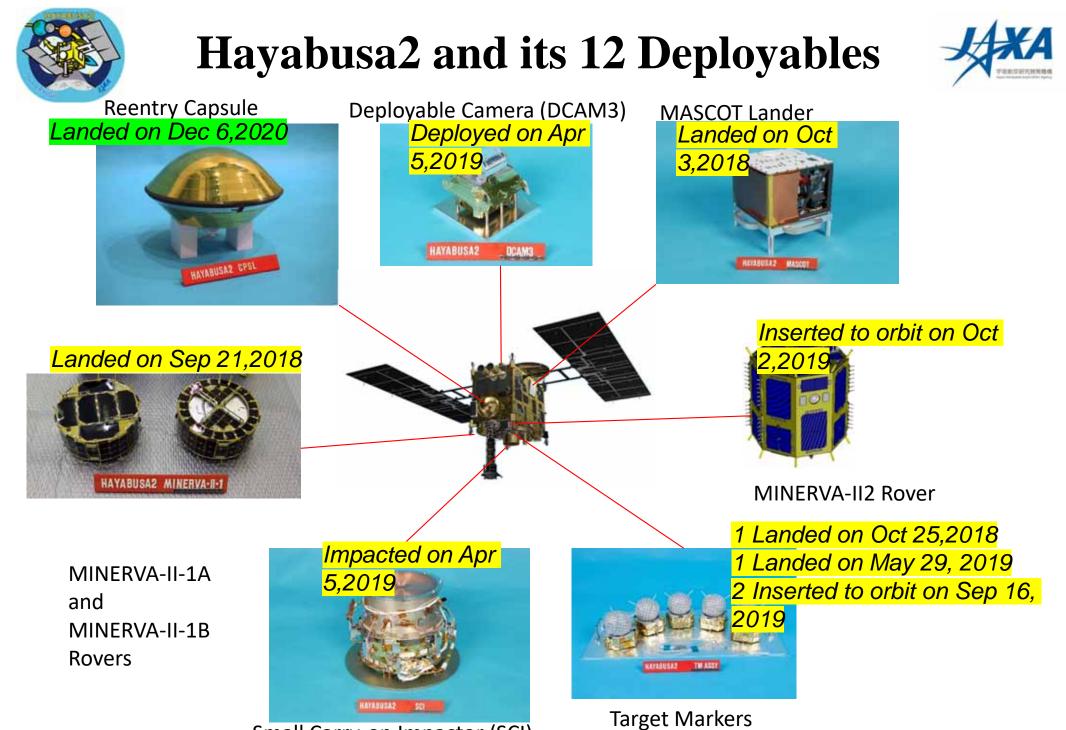
image credit: JAXA

terrestrial celestial body

## Hayabusa2 spacecraft

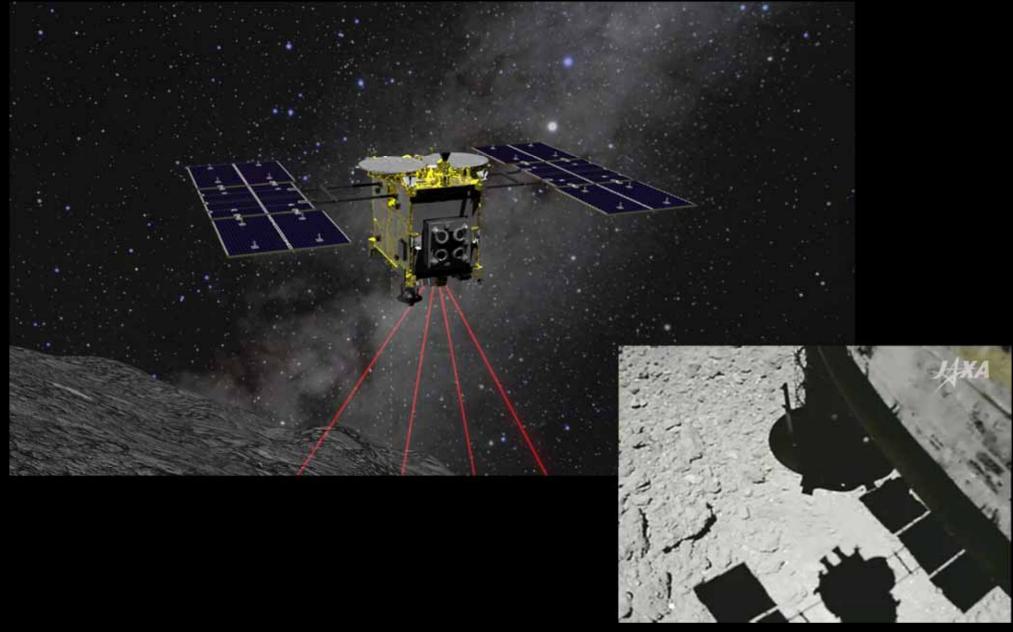






Small Carry-on Impactor (SCI)

### Hayabusa2's biggest highlight: Touch down/Sample Collection

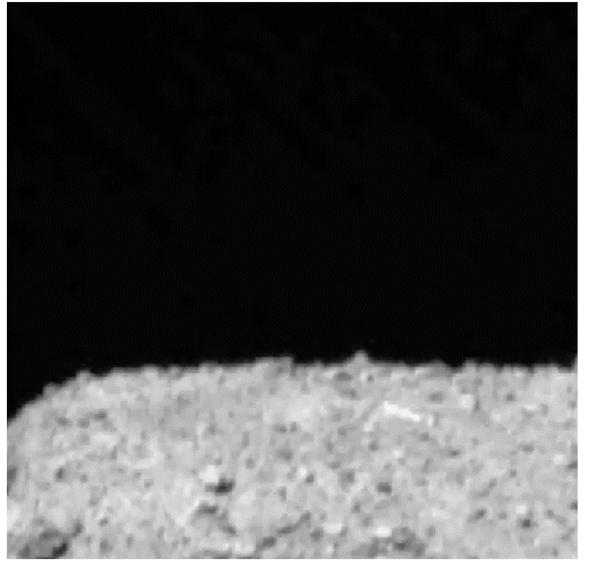




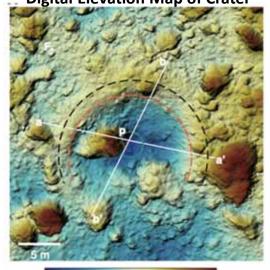
## **Success in making Artificial Crater**



**Cratering Process photographed by DCAM3** 



... Digital Elevation Map of Crater



-2 -1 0 1 2 local height / m

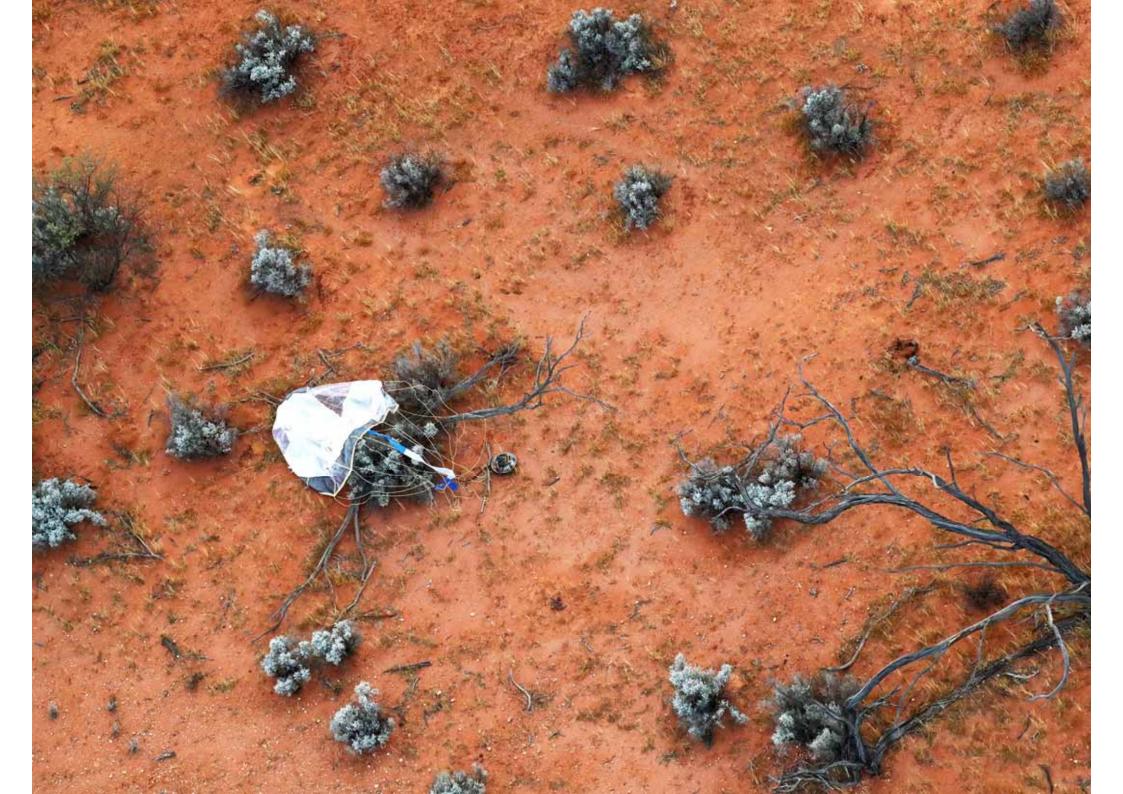
Terrain change before/after impact



(c)Arakawa et al.,2020

Two landings by 1 spacecraft
Access to subsurface material
Landing error only 60cm

Succeeded in the 2nd touchdown on Ryugu at 01:06UTC, July 11, 2019.



#### Ryugu samples found in the sample container! Sample yield : 5.4g (requirement: 0.1g)

## Hayabusa2 Extended Mission

Hayabusa2 (Small Hazardous Asteroid Reconnaissance Probe)
2026 Asteroid "2001 CC21" Flyby
2031 Asteroid "1998 KY26" Rendezvous

1998 KY26

A. Shandie

#### Hayabusa2# From Earth 264 468 321km

398 271 381 04:54:16 15:46:20 0.78au 35.97 km/s

1763 sec

Asteroid Explorer "Hayabusa2"

| Total distance travelled:             | 2      |
|---------------------------------------|--------|
| Time since launch:                    | L+3268 |
| Time since start of Extended Mission: | R+1073 |
| Distance between Sun and spacecraft:  |        |
| Velocity relative to the Sun:         |        |
| Radio wave round trip time:           |        |

# CanSat/CubeSat experience brought me to push the boundary of the space exploration.

## Thank you for your attention.

(C)JAXA

Feb 22, 2019 at Sagamihara Space Operation Center (at the success of the 1st touchdown)