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Microelectronics & Instrumentation Lab



Microélectronique et instrumentation

# Development of 3D Synthetic Vision with network of pico-satellites

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# SATELLITE CONTEXT

- ❖ Now observing the Earth from space is performed by means of a series of satellites.
- ❖ A Pico-satellite constellation plays an important role in missions of vision and surveillance of ground.
- ❖ Spacecrafts may be threatened by the effects of external disturbances.
- ❖ Each satellite networks requires solutions to avoid threats (disturbance).

# Outline

**I. Objective**

**II. Satellites Network**

**III. Choice of Architecture**

**IV. Application**

**V. Conclusion and Outlook**

# OBJECTIVE

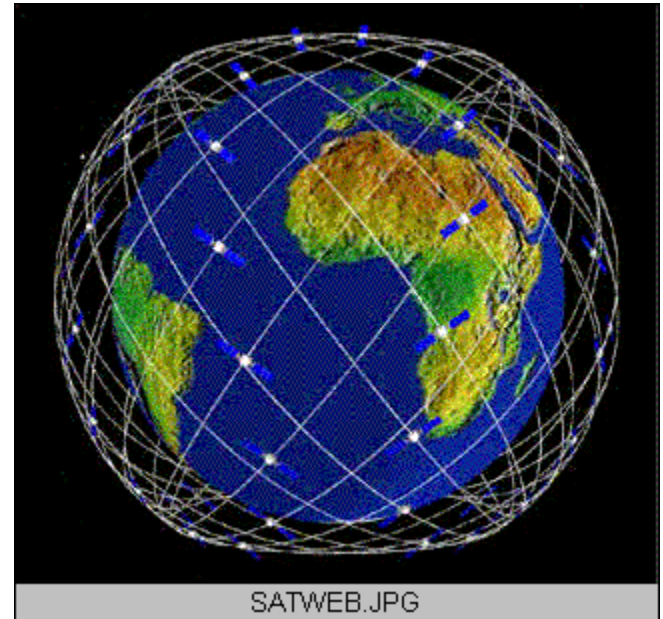
- ❖ Perform an architecture of constellation of Pico-Satellite that meets the coverage of Tunisia:
  - desert region exploitation and monitoring .
  - fire detection, earthquake forecasts and predictions of volcanic activity...
  - Propose optimal solutions that take into account the constraints of the architecture and minimize the number of satellites covering

# SATELLITE NETWORKS

## ❑ Advantages of formation flying

Several identical satellites in cooperative orbits

- Make possible new observing capabilities
- On-orbit reconfiguration within the formations offers multi-mission capability
- Reducing the size and complexity of the satellites in the formation.
- Take advantage of economies of scale
- Can reduce launch costs



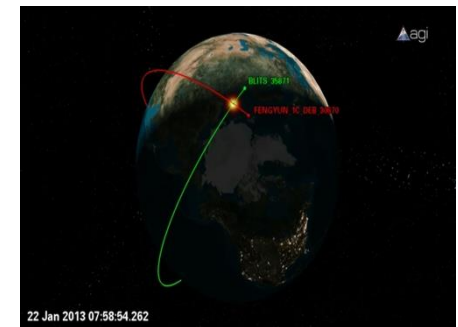
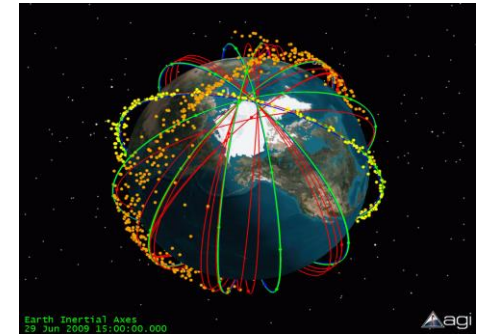
## □ Background Issues

- Two phases of mission of Nanosatellites:
  - ❖ placement on operational orbit.
  - ❖ counter orbital perturbations.

## □ Specific problems in training:

- ❖ Collision of satellites
- ❖ relative motion modeling

Over hundreds of thousands of pieces of this collision build a cloud in LEO

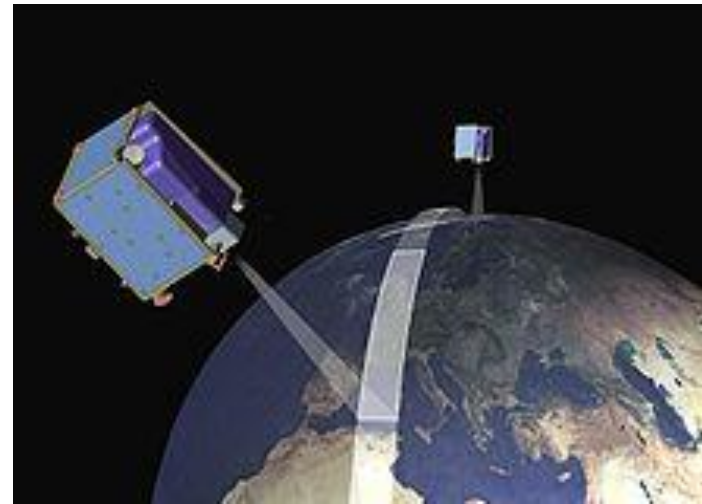
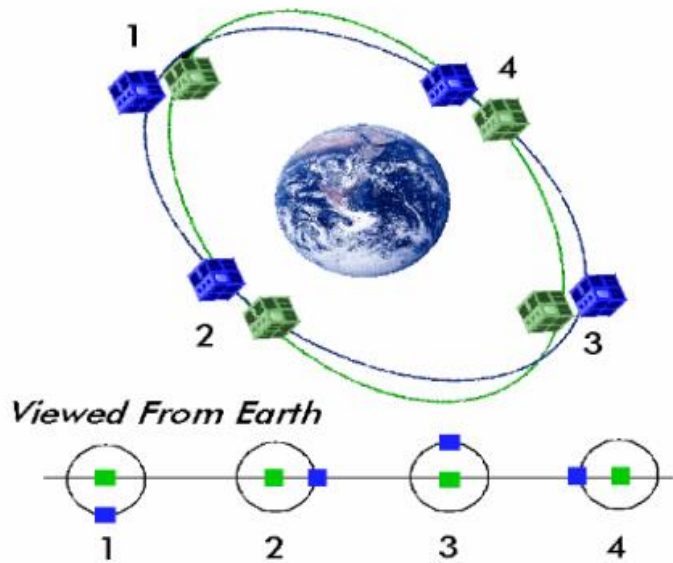


# CHOICE OF ARCHITECTURE

- Architecture avoid any collision.
- Relative motion of each satellite
- Determined and fixed distance between each satellite.
  - Low dimension of training → Differential acceleration = 0
  - Relative Velocity between satellites = 0
  - Stable geometry.

✓ **Undisturbed environment with high accuracy**





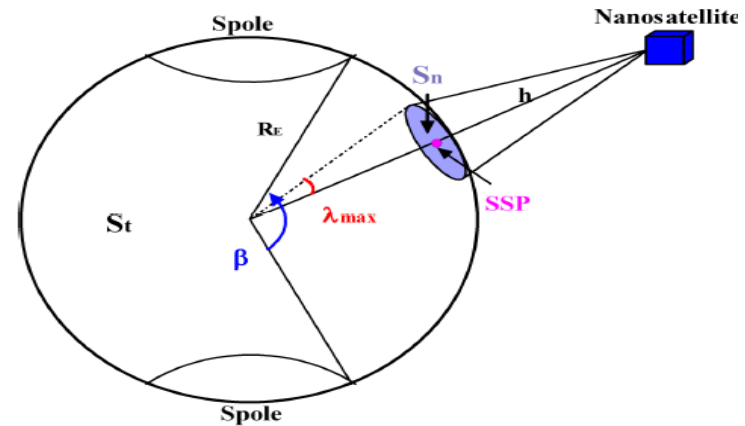
- hierarchy of authority.
- geometry of the formation does not change along the reference orbit.
- suitable for applications in meteorology and environmental.



- ❖ A satellite training must have:
  - As cloakroom, on the other satellite state.
  - A control law that depends on the other member.
- ❖ Their positions are determined by the distance measuring devices DMD and GPS.
  - DMD determine their relative distances to each other
  - GPS determines their position relative to the Earth and checks the distance measured between the satellites.

# APPLICATION

## ❑ Constellation Modules



Design vector variable	$\Sigma$	Range
Constellation altitude	$\sigma_1$	300 Km
Downlink data rate	$\sigma_2$	140 Kbps
elevation angle	$\sigma_3$	$5^\circ \leq \epsilon \leq 10^\circ$

Nano-Satellite for Tunisa Coverage Design Vector

Constant vector variable	$\Gamma$	Value
Latitude	$\gamma_1$	$34^\circ$
Inclination angle	$\gamma_2$	$36^\circ$
$\beta$ angle	$\gamma_3$	$56^\circ$

Nano-Satellite for Tunisa Coverage Constant Vector

# □ orbital parameters



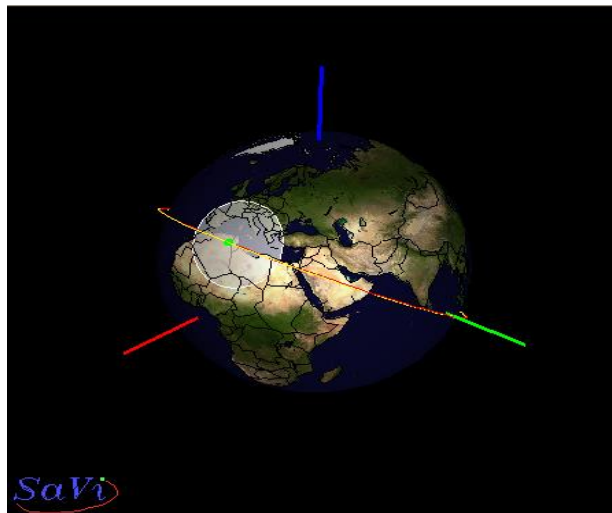
SaVi: Edit satellite parameters (press return to save)

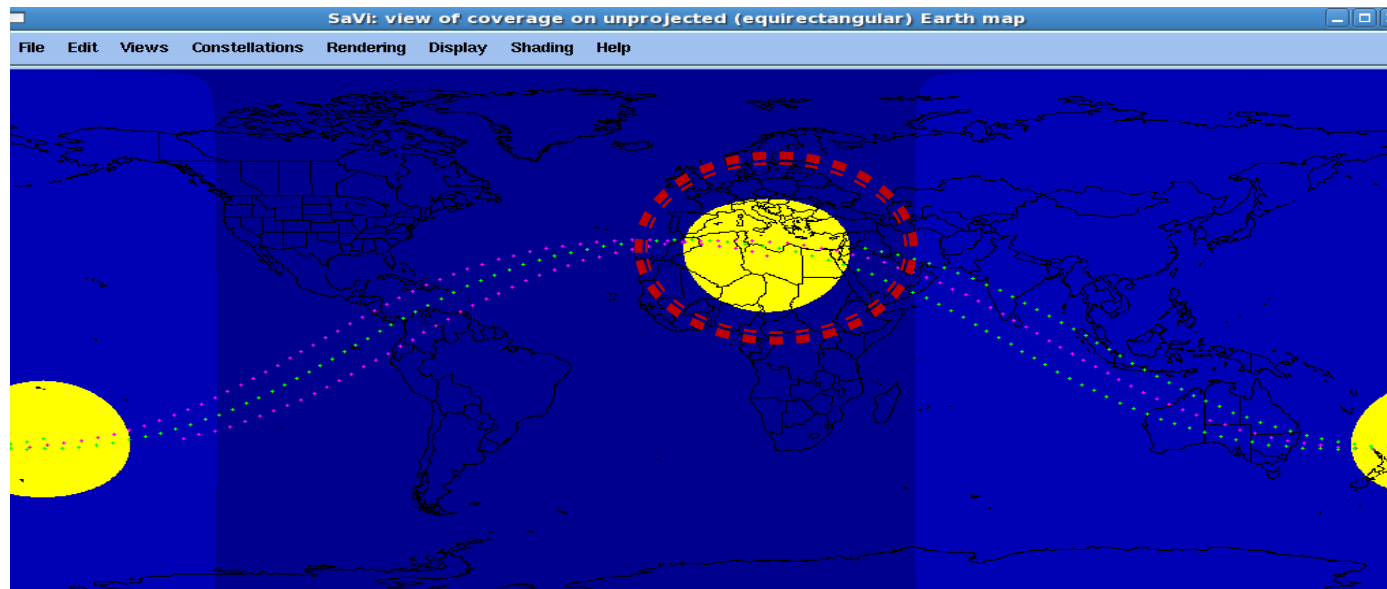
Satellite name  
TUNSAT

Orbital elements		
semi-major axis (km)	6700.00	period (minutes)
eccentricity e	0.0320	-6.7989
inclination (deg)	36.000	9.5491
long. asc. node (deg)	266.400	536.26
arg. of periapsis (deg)	100.800	107.46
time to periapsis (s)	0.000	

☒ Show satellite and groundtrack ☒ Show coverage area ☒ Show orbit

Dismiss window





## Nano-Satellite for Tunisa Coverage

**Terminé!**

Mode de recherche: 2014-11-14 21:58:34      Fuseau horaire UTC +1:00  
 2014-11-15 21:58:34      Passages totaux: 2  
 Lieu: Tunisie (9.5375° E, 36.8869° N)

Satellite poursuivi 1/1: DELTA 2 R/B(1)      Passages: 2

Avance totale: Terminé!

Tri: Terminé!

OK

Number of Nano-satellites	2 Sats
Earth central angle ( $\lambda_{\max}$ )	14,94°
Maximum time in view ( $T_{\max}$ )	7,51 min
Orbital Period	9h
Number of orbit per day	2

## Outputs of Nano-Satellite for Tunisa Coverage

# CONCLUSION AND OUTLOOK

## □ Conclusion

- Review of some problems in LEO satellite networks.
- Find an architecture of satellites network that can interact with the most problems.

## □ Outlook

- Discuss the best architecture of Leader-Follower for the best coverage of the earth (one or two Orbits).
- Sizing coverage area

# Thank You!

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