Results of Group Discussion

Requirements for Attitude Determination and Control System of university satellites

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Will be discussed

- > Sensors for attitude determination
- > Active/passive attitude control
- Configuration of ADCS and its accuracy for different mission types
- ➤ Flexibility of ADCS for onboard reprograming and sensors calibration
- > The place for self-developed ADCS for such kind of projects
- ➤ The possible role of UNISEC and its network for increasing efficiency of university satellite design

Expectations from the discussion

- Defining configurations of ADCS for different types of missions
- Ways of decreasing satellites failure rates for future satellites which are either going to be built by new teams or teams with limited experience (using ADCS example)

Overview of 2 Satellite ADCS

BeeagleSat

Sensors:

Magnetometer

6 Sun sensors (coarse)

1 Sun sensor (precise)

1 Horizon sensor

1 Gyro

Active Control:

3 Magnetorquers

1 Reaction Wheel

Horyu IV

Sensors:

6 Sun sensors (precise)

5 Solar panels

2 Gyros

GPS

Passive Control:

1 Permanent magnet

1 Hysteresis Damper

Minimum Required ADCS components

Remote Sensing ±1 deg

Sensors:

1 Gyro 3-axis

1 Sun sensor (precise) FOV

180 deg

1 Magnetometer 3-axis

Horizon sensor

Active Control:

3 Magnetorquers

3 Reaction Wheel

Scientific Mission ±10 deg

Sensors:

6 Sun sensors (coarse)

1 Magnetometer

Passive Control:

1 Permanent magnet

1 Hysteresis Damper

Group Recommendation

ONLINE PLATFORM

- Basic info for each Spacecraft Subsystem
- History Log of executed missions
- FORUM / FAQ
- Analysis data and Conclusions
- Online Library
- Lessons

Building the Platform

UNISEC members can contribute

- Motivation for volunteers
 - Recognition by UNISEC Community

Networking

Other benefits

Japan Turkey Bulgaria Egypt Mexico Nigeria Ukraine

